

# Flight Vehicle Design VTU CBCS Question Paper Set 2018

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10AE81

**Eighth Semester B.E. Degree Examination, June/July 2017**  
**Flight Vehicle Design**

Time: 3 hrs.

Max. Marks:100

**Note: Answer any FIVE full questions, selecting  
atleast TWO questions from each part.**

**PART – A**

- 1 a. Enlist the ten performance parameter considered for designing an aircraft. (10 Marks)
- b. Consider a typical military Bomber of L/D = 16 warm up and take off fuel fraction is 0.97. Climb fuel fraction is 0.985, cruise R = 1500 n m = 2778 km, C = 0.5/hr, V = 0.6M (same for both the cruise condition 1<sup>st</sup> loiter E = 3 hrs, C = 0.4/hr 2<sup>nd</sup> loiter E = 1/3 hrs. Landing fuel fraction is 0.995. Estimate take off to landing fuel fraction  $W_f/W_0$ . From  $W_f/W_0$ . Calculate the value of  $W_0$ . (10 Marks)

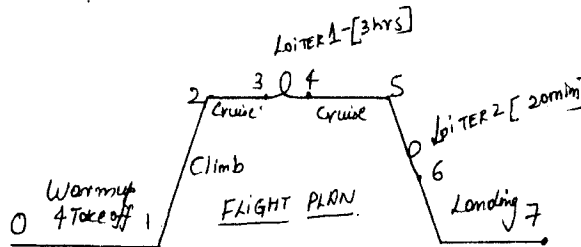


Fig.Q1(b)

- 2 a. Define the term 'Wing Loading'. Briefly explain the consideration for selection of  $\frac{W}{S}$  for an aircraft under design process. (10 Marks)
- b. Derive and explain :
  - i) Wing loading effect on range
  - ii) Effect of aspect ratio on aircraft performance. (10 Marks)
- 3 a. Explain sweep angle selection criteria. (10 Marks)
- b. Show that for a straight, tapered wing, mean aerodynamic chord(MAC) is  $\bar{C} = \frac{2}{3}C_r \left( \frac{\lambda^2 + \lambda + 1}{\lambda + 1} \right)$ , where  $\lambda$  - taper ratio and  $C_r$  is root chord and derive value for  $\Delta_{x/C}$  and  $\Delta_{c/4}$ . (10 Marks)
- 4 a. Show in a graph the variation of drag due to lift, zero lift drag and total drag with velocity. Also show how power required and power available in a piston engine propeller aircraft varies with velocity. (10 Marks)
- b. Explain engine installed thrust correction. (10 Marks)

**PART – B**

- 5 a. Write the equation of motion of landing roll and obtain an expression for landing ground roll distance. (10 Marks)
- b. Explain three common approaches used for active lift enhancement, with the help of neat sketches. (10 Marks)

Important Note - 1 On completing your answers, compulsorily draw diagonal cross lines on the remaining blank pages.

- 6 a. Explain contribution of fuselage using vertical stabilizer towards lateral stability. (10 Marks)  
b. What are neutral point, c.g. margin and static margin? (10 Marks)
- 7 a. Sketch and explain three commonly used landing gear arrangements. (10 Marks)  
b. Explain anti-icing and de-icing systems in an aircraft. (10 Marks)
- 8 a. Explain a typical flight control system. (10 Marks)  
b. Write short note on :  
i) Radio navigation systems  
ii) Aircraft weapon systems. (10 Marks)

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**Eighth Semester B.E. Degree Examination, June/July 2015**

**Flight Vehicle Design**

Time: 3 hrs.

Max. Marks:100

**Note: Answer FIVE full questions, selecting at least TWO questions from each part.**

**PART – A**

- 1 a. Explain overview of airplane design process. (08 Marks)  
b. Calculate near exact weight for an airplane from a guess value for the following data, where  $W_e$  is the empty weight and  $W_o$  is the takeoff weight.  
$$\frac{W_e}{W_o} = 0.97 W_o^{-0.05} \text{ and } W_o = \frac{10,500}{1 - 0.4 - \frac{W_e}{W_o}}$$
 (12 Marks)
- 2 Explain the effect of wing loading on stall speed, take off distance, catapult take – off, landing distance, cruise, loiter for endurance, instantaneous turn and sustain turn. (20 Marks)
- 3 a. Explain the wing sweep angle selection criteria? (10 Marks)  
b. Draw layout of a spread sheet for wing design. (10 Marks)
- 4 a. What are engine installed thrust correction? (10 Marks)  
b. A jet engine performance data is given below :  
rpm = 9500  
EGT = 450°C  
 $w_f$  (fuel consumption) = 1830 Kg/hr  
 $w_a$  = (air consumption) = 91 Kg/Sec  
 $F_n$  (net thrust) = 4510 Kg  
TFSC (thrust specific fuel consumption) = 0.5  
The test is carried out at pressure of 102.6 kPa and ambient temperature of 30° C. Correct the test data for ISA conditions (pressure 101.3 kPa and temperature 15° C) (10 Marks)

**PART – B**

- 5 a. What is balanced field length? (06 Marks)  
b. Draw spread sheet layout for take – off and landing distance. (14 Marks)
- 6 a. Explain rudder area sizing. (12 Marks)  
b. What is neutral point, c.g. margin and static margin? (08 Marks)
- 7 a. Explain Alternating current electrical power system for an aircraft. (10 Marks)  
b. What is Castoring – wheel geometry? (10 Marks)
- 8 a. Explain a typical flight control system. (12 Marks)  
b. Briefly describe weapon carriage and gun installation on military aircraft. (08 Marks)

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Important Note : 1. On completing your answers, compulsorily draw diagonal cross lines on the remaining blank pages.  
2. Any revealing of identification, appeal to evaluator and /or equations written eg. 42+8 = 50, will be treated as malpractice.

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**Eighth Semester B.E. Degree Examination, June/July 2016**  
**Flight Vehicle Design**

Time: 3 hrs.

Max. Marks: 100

**Note: Answer any FIVE full questions, selecting  
atleast TWO questions from each part.**

**PART – A**

- 1 a. Define Design. Briefly explain the design process of an aircraft with a flow chart. (10 Marks)  
 b. Mention the performance parameter for designing an aircraft. (10 Marks)
- 2 a. Derive an expression for wing loading effect on Glide rate. (10 Marks)  
 b. Define Load Factor. Show that the designer can minimize  $\frac{D}{W}$  with respect to  $\frac{W}{S}$  with the  
 relation  $\frac{W}{S} = \frac{q}{n} C_{\left(\frac{L}{D}\right)_{\min}}$  (10 Marks)
- 3 a. Write short notes on the following :  
 (i) Airfoil shape selection  
 (ii) Base Drag Estimation. (10 Marks)  
 b. Mention and briefly explain the volume consideration mode while designing an aircraft fuselage. (10 Marks)
- 4 a. Discuss the following topics in detail:  
 (i) Propulsion selection  
 (ii) Propeller Design for cruise condition (10 Marks)  
 b. Explain the spread shut approach for Turbo-Jet engine sizing. (10 Marks)

**PART – B**

- 5 a. Derive an expression for aircraft ground roll. (10 Marks)  
 b. Enlist all phases of flight landing with schematic sketch and mention all the expression related to each phase. (10 Marks)
- 6 a. Discuss briefly any two refined weight estimation methods used in aircraft. (10 Marks)  
 b. Explain longitudinal stability effect on performance of the aircraft. (10 Marks)
- 7 a. Sketch and explain three commonly used landing gear arrangements. (10 Marks)  
 b. Explain anti icing and de-icing system in an aircraft. (10 Marks)
- 8 a. Explain a typical flight control system. (10 Marks)  
 b. Write short notes on :  
 (i) Radio navigation system  
 (ii) Aircraft weapon system. (10 Marks)

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